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NPIC/R-865/64 October 1964 Сору

PHOTOGRAPHIC INTERPRETATION REPORT

11 Pages

LAUNCH COMPLEX D

AND WINGED VEHICLE AT LAUNCH AREA 3D

KAPUSTIN YAR/VLADIMIROVKA

MISSILE TEST CENTER, USSR

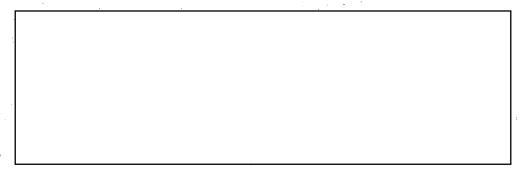
DECLASS REVIEW by NIMA/DOD

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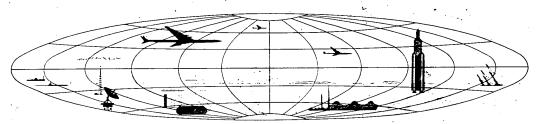




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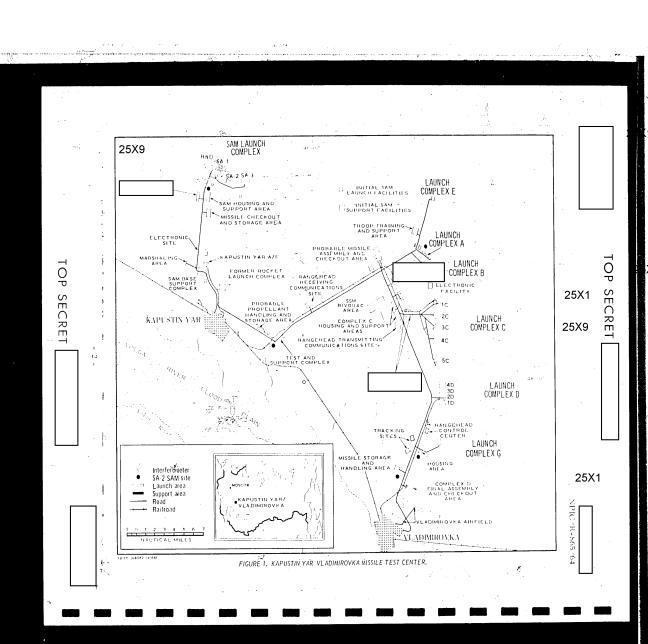


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GROUP I Excluded from automatic dewngrading and declassification

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	LAUNCH (	COMPLEX D	AND WING	ED VEHICLE AT I	.AUNCH	AREA 3D,
•	KAPU	STIN YAR/	VLADIMIROVKA	MISSILE TEST	CENTER,	USSR
	This photogr prepared in res P0432 R45-64, U 81,587, and GMA of reports upda facilities at the K sile Test Center	ponse to requestions of the policy of the po	XX 64-14, C-S14- is one of a series ion on the SSM adimirovka Mis-	a scarred area north launch pad. The re- over a permanent s the rear of the laur some manner. This as it is connected by control section of t	ear of the wmall object, nicher is fas is probable co	inged vehicle is suggesting that stened to this in y a launch point induit back to the
25X1D		covérs Launch includes a de:	Complex D (Fig- ailed study of a	structures on the pation in ently have been remthe pad is a slightle	oved and all y raised pl	appar- that remains on atform or line.
25X1D		provided pled vehicle on a	hotographic cov- an inclined prob- a 3D, KY/VMIC	There were als vehicles/pieces of e the two roads enteribly and checkout built position. The poor does not permit at mensuration of these	quipment or ng the drive ding to the r resolution on ny further	-through assem- ear of the launch f the photography
1 1	fiable, is on pho quality and all c as approximate. slant length of a delta wing at th wing is visible,	tegraphic cove dimensions mu The winged vo approximately te rear. Only so an estimat	rage of only fair est be considered shicle has a total. 70 feet and has a one side of this ted total width of ate as the center	because of the lar photography. One identified as a revet	O (Figure 4) arent change ough two ite raphy canno ger scale of these i ment, can n	ms identified on whe reevaluated of tems, originally town be seen to be
25X1D	point on the 5- estimated. This the front. Appro there are possib or measurement items. The del a heavy fairing of joins the fusela from the tail. The a sweep back The winged	to 10-foot "fus fuselage appoximately 15 follows can be dete to wing appear or double delta ge approximate the one leading of approximate vehicle is incline.	selage" must be ears to taper (at set from the nose. No configuration rmined for these as to have either section where it ely 30 to 35 feet edge visible has	probable cargo truc launch pad and an approximately 70 fe line immediately b served launch struc	the second and the se	photography, eway leading be- ch Area 1D is a on the Site 1D-3 y shaped object ocated on the rail
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in desired, restorable improblementation of the months and the many of the second of t

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25X1D	25X1D NPIC/R-865 64	
iameter just southeast of the drive-through uilding: Another building previously uniden- fied as to function can now be seen to be a eat plant.  Launch Area 4D (Figure 8) has shown no	All construction efforts had apparently ceased before and the partially constructed facilitiesthe pad area,	2 2
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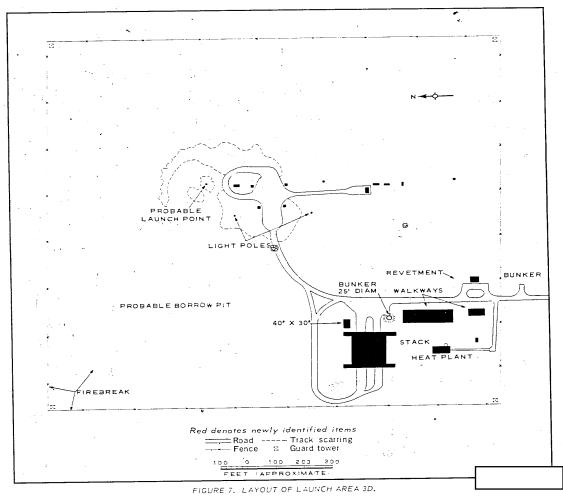
25X1D

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area--show no change on the photography. The launch area still is fence secured but a gap is evident in the fenceline adjacent to the gate across the access road. The launch pad appears unserviceable but there is ample

area to either side which could be utilized for field exercises or firings.

The Final Assembly and Checkout Area (Figure 9) is located approximately 8 nautical miles south-southwest of Launch Complex D



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